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### The Millimeter-Wavelength Sulfur Dioxide Absorption Spectra Measured Under Simulated Venus Conditions

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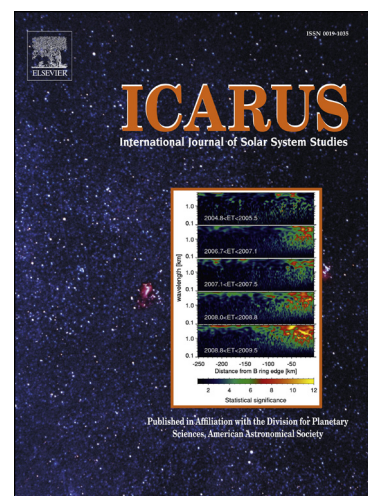
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1 The Millimeter-Wavelength Sulfur Dioxide Absorption  
2 Spectra Measured Under Simulated Venus Conditions

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7 **Abstract**

Over 130 laboratory measurements of the 2-4 millimeter wavelength opacity of sulfur dioxide in a carbon dioxide atmosphere under simulated conditions for the upper Venus troposphere (temperatures between 308-343 K and pressures between 0.03- 2 bar) have been made. These measurements along with the centimeter wavelength measurements by Steffes et al. (2015) have been used to empirically assess existing formalisms for sulfur dioxide opacity in a carbon dioxide atmosphere (Fahd and Steffes (1992) and Suleiman et al. (1996)). The Van Vleck and Weisskopf Model (VVW) used by Fahd and Steffes with the JPL rotational line catalog (Pickett et al. (1998)) was found to fit 85.88% of all 500 measurements within the 2-sigma uncertainty. This work will improve the confidence in retrievals of the atmospheric abundance of sulfur dioxide from millimeter-wavelength observations of the Venus atmosphere.

8 *Keywords:* Venus, atmosphere, Radio observations, Atmospheres,  
9 composition, Spectroscopy

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## 10 1. Introduction

11 Active and passive microwave remote sensing techniques have been ex-  
12 tensively used in the study of our sister planet, Venus. Unlike Earth's atmo-  
13 sphere, the Venus atmosphere is mostly comprised of gaseous carbon dioxide  
14 ( $\text{CO}_2$ ).  $\text{CO}_2$  comprises 96.5% of the atmosphere along with gaseous nitrogen  
15 ( $\text{N}_2$ ) at about 3.5% (Oyama et al., 1980). The Venus atmosphere has multi-  
16 ple trace constituents such as sulfur dioxide ( $\text{SO}_2$ ), carbon monoxide (CO),  
17 water vapor ( $\text{H}_2\text{O}$ ), carbonyl sulfide (OCS), and sulfuric acid vapor ( $\text{H}_2\text{SO}_4$ )  
18 (Jenkins et al., 1994; Krasnopolsky and Pollack, 1994).

19 While the collision induced absorption of  $\text{CO}_2$  dominates the continuum  
20 millimeter wave emission spectrum at Venus, spatial variations in the emis-  
21 sion are attributable to cloud related gases such as  $\text{SO}_2$ ,  $\text{H}_2\text{SO}_4$  (g), and  
22  $\text{H}_2\text{SO}_4$  condensates (Steffes and Eshleman, 1981; Fahd and Steffes, 1992;  
23 Steffes, 1985). At higher pressures  $\text{H}_2\text{SO}_4$  thermally dissociates, forming  
24  $\text{H}_2\text{O}$  and  $\text{SO}_3$ , both of which exhibit relatively small amounts of microwave  
25 absorption at the abundance levels present in the Venus atmosphere (Ho  
26 et al., 1966; Steffes and Eshleman, 1981).

27 With the completion of the ESA Venus Express Mission, earth-based cen-  
28 timeter and millimeter-wavelength observations of Venus are becoming more  
29 important. The improved sensitivity of the NRAO EVLA (Extended Very  
30 Large Array) and the high resolution of ALMA (Atacama Large Millime-  
31 ter Array) provide new capabilities for studying the atmosphere of Venus.  
32 Knowledge of the absorption properties of sulfur dioxide is important in the  
33 analysis of data from these earth-based observations.

34 Utilizing the millimeter-wavelength system at the Planetary Atmospheres

35 Laboratory at Georgia Institute of Technology, it has been possible to simu-  
36 late the upper troposphere of Venus and take precision measurements of the  
37 millimeter-wavelength properties of sulfur dioxide. Using the measurements,  
38 a model that accurately predicts the opacity of sulfur dioxide in the Venus  
39 atmosphere has been verified. Applying this opacity model to a radiative  
40 transfer model will make it possible to determine the source of variations in  
41 the Venus millimeter-wavelength emission, such as were observed by de Pater  
42 et al. (1991) and Sagawa (2008).

43 Sagawa (2008) attributes the Venus millimeter-wavelength continuum  
44 brightness temperature variations to spatial variations in the abundances  
45 of both gaseous  $\text{H}_2\text{SO}_4$  and  $\text{SO}_2$  just below the cloud layer (48 km alti-  
46 tude). Sagawa (2008) has also suggested that the effects of both constituents  
47 can be distinguished based on differences in frequency dependencies of their  
48 millimeter-wavelength opacities. However, to accomplish this, high accuracy  
49 models must be developed that characterize the opacity of each constituent  
50 and their frequency dependence.

51 Observing how  $\text{SO}_2$  varies spatially in the Venus atmosphere allows for  
52 a better understanding of the atmospheric dynamics that occur on Venus  
53 (Encrenaz et al., 2012, 2013; Sandor et al., 2010). Detailed thermochemical  
54 modeling indicates that the probable source of  $\text{SO}_2$  in the lower atmosphere  
55 of Venus is due to volcanic activity, even though no volcanic activity has  
56 been directly observed (Esposito et al., 1988; Mahieux et al., 2014).

57 Radio observations of Venus in the 2-4 millimeter-wavelength range have  
58 the potential to allow for detection of variations in the abundance of  $\text{SO}_2$  in  
59 the 35-70 km altitude range which includes the cloud layers. The altitude

60 range of the continuum emission in this wavelength range is constrained by  
61 the dominant opacity of CO<sub>2</sub>. However localized variations in the emission  
62 can be attributed to localized abundance variations in cloud-related gases  
63 such as SO<sub>2</sub>. Recently, observations of Venus with the Nobeyama millimeter-  
64 wave array of the 103 GHz continuum emission showed a 25% variation in  
65 brightness with position on the disk (Sagawa, 2008). While 1.3 and 2.0 cm  
66 continuum emissions from Venus indicated dark ( $\sim 3\%$ ) polar regions con-  
67 sistent with increased H<sub>2</sub>SO<sub>4</sub> vapor abundance due to vaporization of cloud  
68 condensate from the downwelling characteristic of Hadley cell circulation  
69 (Jenkins et al., 2002), the 3 mm maps show much stronger variations over a  
70 range of different locations, with some indication of diurnal variation. de Pa-  
71 ter et al. (1991) reports significant variations (10%) in the 2.6 mm emission  
72 maps of Venus made with the Hat Creek Interferometer.

73 It is well understood that the microwave emission spectrum of Venus re-  
74 flects the abundance and distribution of its constituents. A critical limiting  
75 factor in sensing these constituents is the knowledge of their microwave ab-  
76 sorption properties under Venus conditions. The millimeter-wavelength ab-  
77 sorption of SO<sub>2</sub> at 94.1 GHz has been measured by Fahd and Steffes (1992).  
78 Using newer technology it is possible to measure a wider spectrum with  
79 higher precision. Improved laboratory capabilities also allow a wider range  
80 of environmental conditions, similar to those actually being probed, to be  
81 simulated. The millimeter-wavelength system used in these experiments is  
82 able to reproduce conditions similar to those that exist on Venus. These  
83 results, in addition to centimeter-wave absorption spectra already measured  
84 by Steffes et al. (2015), have been used to help choose a model that best rep-

85 represents the centimeter- and millimeter-wavelength opacity of SO<sub>2</sub> in a CO<sub>2</sub>  
86 atmosphere.

## 87 2. Measurement Theory and System

88 Verifying the millimeter-wavelength absorption spectrum of SO<sub>2</sub> is im-  
89 portant for the study of the atmosphere of Venus. Making measurements  
90 under simulated Venus conditions assures the accuracy of any model derived  
91 from such measurements.

92 In this experimental program, the quality factor ( $Q$ ) of a resonant mode  
93 of a resonator is used to measure the absorption of a gas or gas mixture at  
94 that resonant frequency (Hanley and Steffes, 2007). The relationship between  
95 quality factor and absorption is given by

$$\alpha = 8.686 \frac{\pi}{\lambda} \left( \frac{1 - \sqrt{t_{loaded}}}{Q_{loaded}^m} - \frac{1 - \sqrt{t_{matched}}}{Q_{matched}^m} \right) \text{ dB/km} \quad (1)$$

96 where  $\alpha$  is the absorptivity of the gas,  $\lambda$  is the wavelength of the resonance  
97 (in km),  $t$  is the transmissivity of the resonance, and  $Q$  is the quality factor of  
98 the resonance. The subscripts *loaded* and *matched* refer the measurements  
99 made with the test gas and pure CO<sub>2</sub> respectively. The full derivation of this  
100 can be found in Appendix A.

101 Described below is the laboratory equipment and measurement procedure  
102 used in the measurement of the millimeter-wavelength absorptivity of gaseous  
103 SO<sub>2</sub> under simulated Venus conditions.

### 104 2.1. Millimeter-Wavelength Measurement System

105 The high-sensitivity millimeter-wavelength system used for measuring the  
106 opacity of gaseous sulfur dioxide under Venus conditions is similar to the one

107 used by Devaraj and Steffes (2011). The system is comprised of two subsys-  
108 tems for measuring different bands of the millimeter-wavelength spectrum  
109 (W-band/F-band). The simulator consists of a glass pressure chamber and  
110 a stable temperature chamber. Experiments using this system were done at  
111 pressures as high as 2 bars and temperatures up to 350 K. The W-band sub-  
112 system is used for measurements in the 2.7-4 millimeter-wavelength range  
113 while the F-band system is used for the 2-3 millimeter-wavelength range.  
114 Multiple resonances are measured in each band so as to develop an absorp-  
115 tivity spectrum. There are no resonances on the edge of the spectral range  
116 of each system; therefore no edge effects are present. The uncertainties as-  
117 sociated with measurements using this system are described in Appendix  
118 B.

#### 119 *2.1.1. W-band Subsystem*

120 The W-band measurement system is used to measure the 2.7-4 mm-  
121 wavelength properties of sulfur dioxide and shown in Figure 1.

122 A synthesized swept signal generator (HP 83650B) is used to generate a  
123 signal in the 12.5-18.3 GHz range which is fed to a times-six active multiplier  
124 chain (AMC) via low-loss, high frequency coaxial cables. The active multi-  
125 plier then feeds the 75-110 GHz signals (swept over the range covered by each  
126 single resonance) to the Fabry-Perot resonator via WR-10 waveguides. The  
127 millimeter-wavelength radio frequency (RF) signal from the output port of  
128 the Fabry-Perot resonator (FPR) is fed via waveguide to a QuinStar Technol-  
129 ogy QMH series harmonic mixer. The local oscillator (LO) and the interme-  
130 diate frequency (IF) are connected via an external diplexer. The harmonic  
131 mixer is locked to the 18th harmonic of the spectrum analyzer LO and is

132 used in the “external mixer” mode with the spectrum analyzer (HP 8564E).

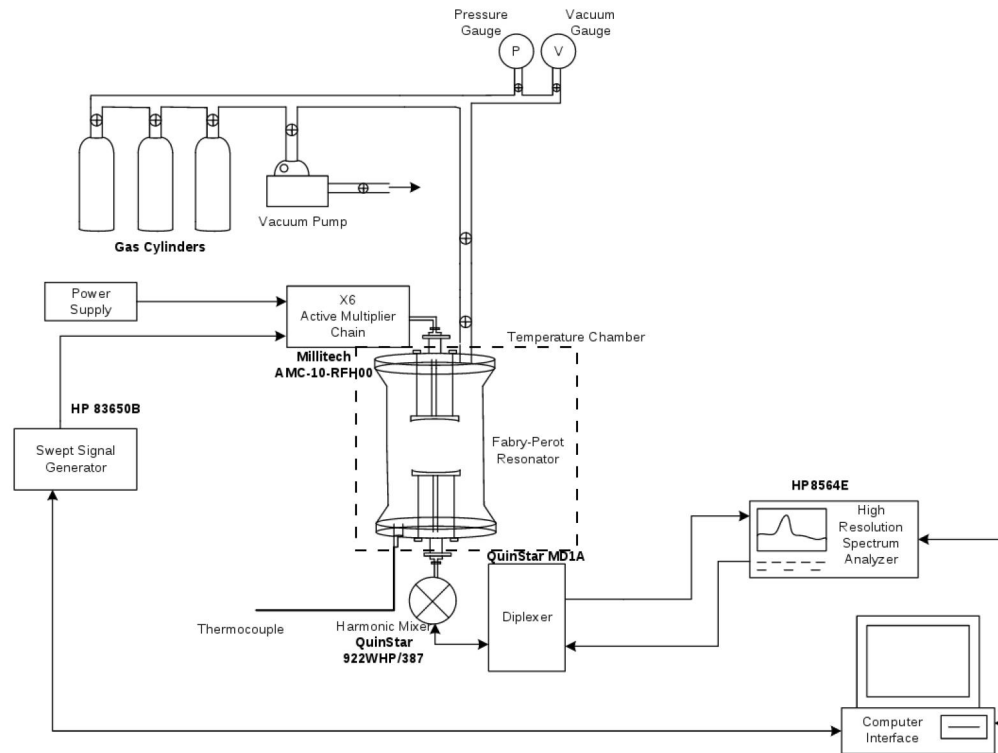


Figure 1: Block diagram of the W-band measurement system. Solid lines represent the electrical connections and the arrows show the direction of the signal propagation. Valves controlling the flow of gasses are shown by small crossed circles.

### 133 2.1.2. F-band Subsystem

134 The F-band measurement system is used to measure the 2-3 mm-wavelength  
135 properties of sulfur dioxide and is shown in Figure 2.

136 The swept signal generator (HP 83650B) is used to generate a signal in  
137 the 33-50 GHz range which is amplified and fed through a frequency tripler.  
138 The output of the tripler is fed to the input of the FPR via WR-8 waveguides.



139 The RF signal from the output port of the FPR is fed to a harmonic mixer  
 140 which can operate with an LO frequency as high as 18 GHz. An external  
 141 diplexer is used to combine the IF and LO signals. For a particular RF and  
 142 IF frequency, the LO frequency can be computed using

$$f_{LO} = \frac{f_{RF} - f_{IF}}{N_H} \quad (2)$$

143 where  $N_H$  is the lowest integer such that  $f_{LO} < 18GHz$ .

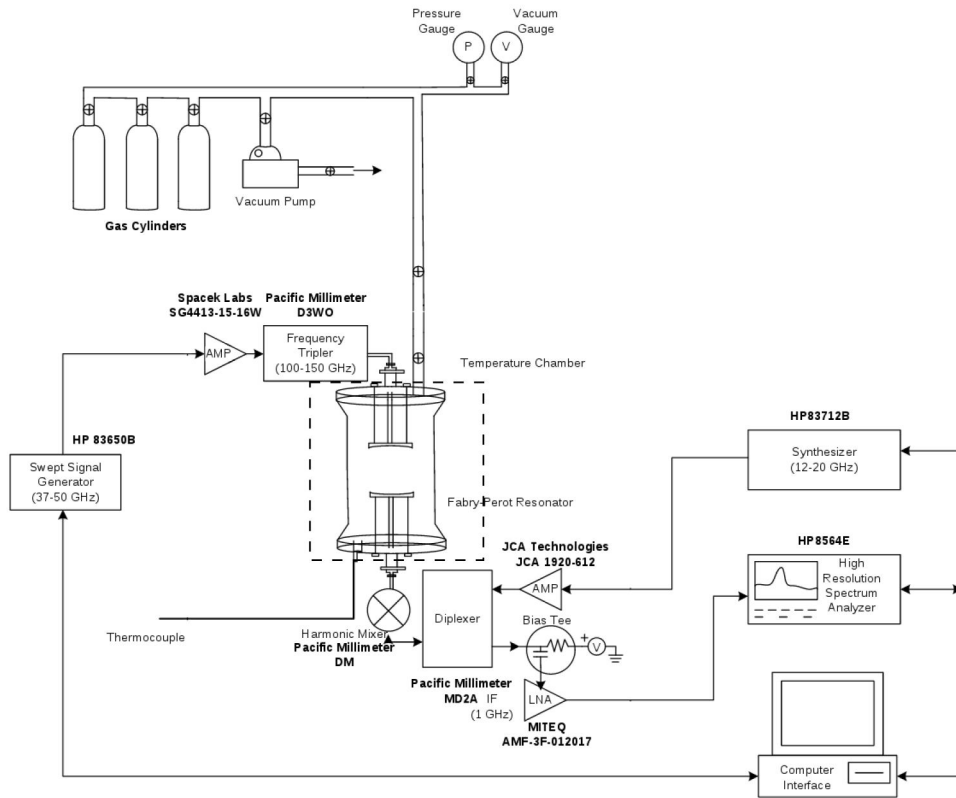


Figure 2: Block diagram of the F-band measurement system. Solid lines represent the electrical connections and the arrows show the direction of the signal propagation. Valves controlling the flow of gasses are shown by small crossed circles.

### 144 *2.1.3. Data Handling Subsystem*

145 The data acquisition system consists of a computer connected to the  
146 spectrum analyzer (HP 8564E), swept signal generator (HP 83650B), and  
147 continuous wave (CW) signal generator (HP 83712B, the local oscillator for  
148 the F-Band system) via a general purpose interface bus (GPIB). The instru-  
149 ments are controlled via Matlab script and their appropriate programming  
150 language. The software used is similar to that described in Devaraj and  
151 Steffes (2011) with modifications for equipment changes.

### 152 *2.2. Measurement Procedure*

153 The most important prerequisite for performing measurement of gas prop-  
154 erties is ensuring a leak-proof system. Pressure integrity was verified using  
155 two methods: the first is by drawing a vacuum inside the FPR and verifying  
156 the integrity of the vacuum over time. The second is by adding a positive  
157 pressure of CO<sub>2</sub> to the system and making sure there are no leaks in any of  
158 the connectors and valves. Ensuring a leak-proof system allows for not only  
159 precise measurements but also ensures no toxic gases are released into the  
160 testing environment.

161 After the system is ensured to be leak-proof and at a stable temperature,  
162 a vacuum is drawn and a measurement is taken using the appropriate subsys-  
163 tem (W-band for 2.7-4 mm-wavelengths, F-band for 2-3 mm-wavelengths).  
164 This allows for a baseline measurement of the FPR's resonances and the  
165 quality factor. Once this baseline is established the gas under test is added  
166 to the system.

167 Once the gas temperature has stabilized, another set of tests measuring  
168 the resonant frequencies along with the quality factors is taken. More gas is

169 added and the procedure is repeated until measurements at all suitable pres-  
170 sures are taken. A vacuum is drawn once again but this time it is pumped  
171 overnight due to the possibility of adsorption (or “sticking”) of the gas be-  
172 ing tested ( $\text{SO}_2$ ) to metal surfaces inside the vessel. This second vacuum  
173 measurement is taken to measure any possible system drift.

174 Once the second vacuum measurement is taken,  $\text{CO}_2$  is then added to  
175 the chamber until the resonances are matched to the same frequency as the  
176 test gas (note that at the pressures and frequencies used for our experiment,  
177 pure  $\text{CO}_2$  is essentially lossless). Again measurements are taken and this  
178 is repeated for every pressure of the test gas. Once completed a vacuum is  
179 again drawn and another test is taken.

180 Lastly the system is set up for a transmissivity test where we measure  $t$   
181 (equation A.5) for each given resonant frequency. This is done by bypassing  
182 the Fabry-Perot resonator and connecting the input and output waveguides  
183 through a WR-10 20 dB directional coupler. The signal level is then measured  
184 and used to calculate the transmissivity,  $t$ . The system is then set back up  
185 and is ready for a new test.

### 186 3. Results

187 High accuracy laboratory measurements of the temperature and pressure  
188 dependence of the millimeter-wavelength absorption of gaseous  $\text{SO}_2$  in a  $\text{CO}_2$   
189 atmosphere have been conducted at 308 K and 345 K and at pressures from  
190 30 mbar to 3 bars for wavelengths between 2-4 millimeters (Table 1). Fig-  
191 ures 3–8 show the typical results of these absorptivity measurements with  
192 the accompanying  $2\sigma$  uncertainties. For comparison purposes, these plots

193 also show two known formalisms of SO<sub>2</sub>'s absorptivity. One developed by  
194 Suleiman et al. (1996) and the second by Fahd and Steffes (1992) but using  
195 the new JPL line catalog (Pickett et al., 1998).

196 The Suleiman et al. (1996) model employs a Ben-Reuven lineshape which  
197 is the quantum mechanical equation which treats collisional broadening as  
198 collisions on a classical oscillator. The formalism employs pressure-broadened  
199 line widths, coupling elements, and pressure shift terms (Waters, 1976). The  
200 Fahd and Steffes (1992) formalism uses a traditional Van Vleck-Weisskopf  
201 lineshape which assumes that the collision-induced phase shifts and momentum-  
202 reversing collisions are random (Waters, 1976) and thus no coupling elements  
203 or pressure-shift terms are required.

204 Since each resonance has a different quality factor ( $Q$ ), each spectral data  
205 point will have a different uncertainty. A low transmissivity,  $t$ , for a specific  
206 resonance will result in a low signal to noise ratio, which in turn affects the  
207 sensitivity of each resonance. The test gas can become opaque enough such  
208 that the resonance is no longer detectable. When this happens the resonance  
209 is not plotted or used in the model analysis.

Table 1: Testing matrix for  $\text{SO}_2$ 's microwave absorption properties at 2-4 millimeter-wavelength.

Temperature	$\text{SO}_2$ Pressure	$\text{CO}_2$ Pressure	Subsystem
308 K	110 mbar	0, 0.83, 1.87 bar	W-band
343 K	85 mbar	0, 0.84, 1.92 bar	W-band
343 K	51 mbar	0, 0.87, 1.94 bar	W-band
343 K	76 mbar	0, 0.83, 1.87 bar	F-band
343 K	28 mbar	0, 0.91, 1.97 bar	F-band
308 K	25 mbar	0, 0.97, 2.00 bar	F-band

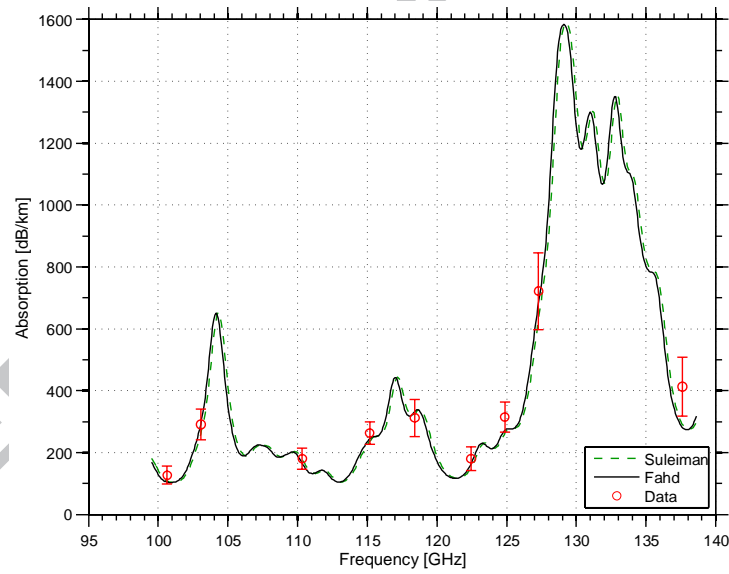


Figure 3: Opacity data using the 2-3 mm-wavelength system for a mixture of 76 mbars of  $\text{SO}_2$  and 14 mbars of  $\text{N}_2$  at a temperature of 344.4 K compared to various models

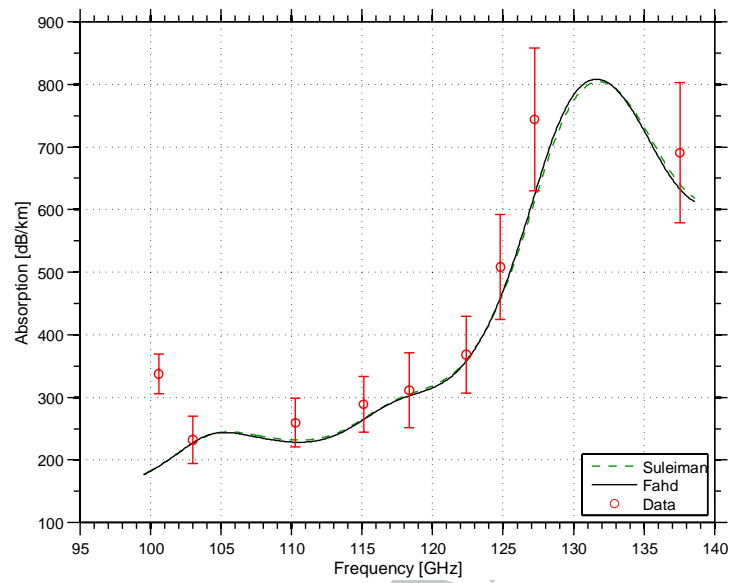


Figure 4: Opacity data using the 2-3 mm-wavelength system for a mixture of 76 mbars of  $\text{SO}_2$  and 833 mbars of  $\text{CO}_2$  at a temperature of 344.6 K compared to various models

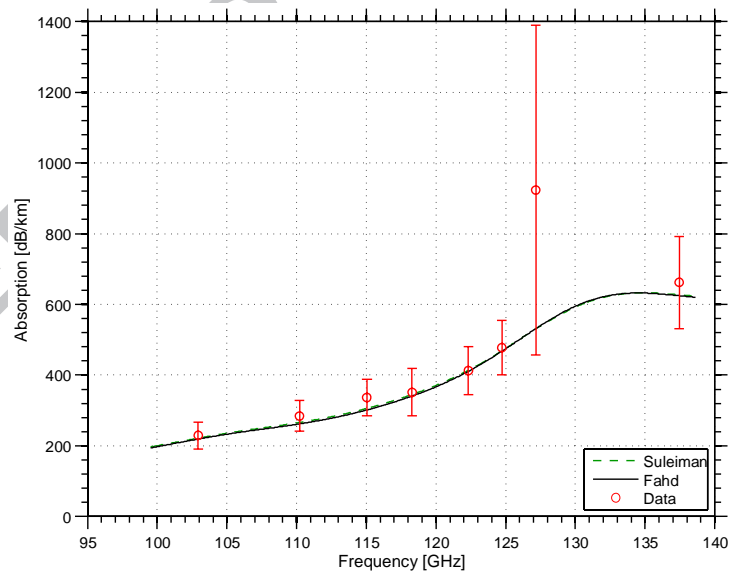


Figure 5: Opacity data using the 2-3 mm-wavelength system for a mixture of 76 mbars of  $\text{SO}_2$  and 1.877 bars of  $\text{CO}_2$  at a temperature of 343.9 K compared to various models

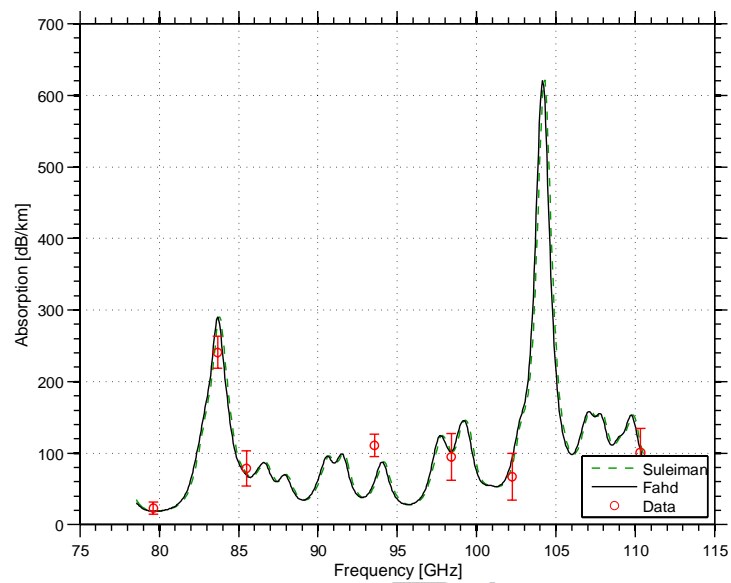


Figure 6: Opacity data using the 2.7-4 mm-wavelength system for a mixture of 51 SO<sub>2</sub> and 9 mbars of N<sub>2</sub> at a temperature of 343.1 K compared to various models

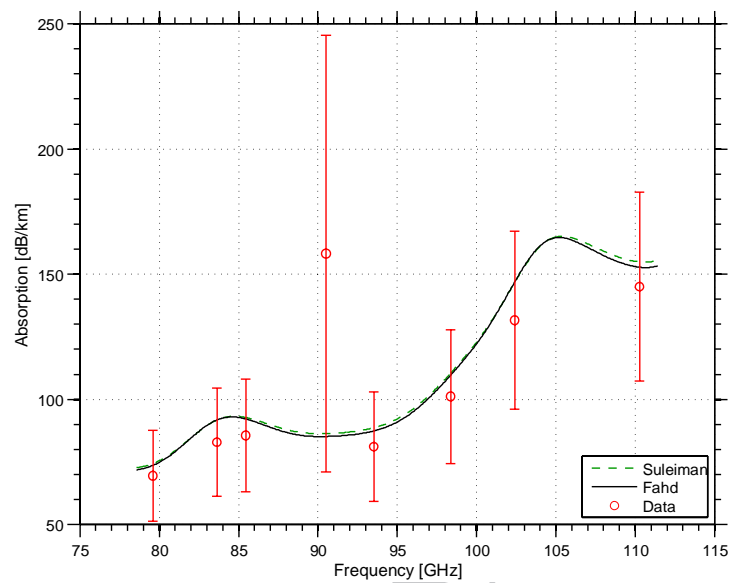


Figure 7: Opacity data using the 2.7-4 mm-wavelength system for a mixture of 51 mbars of  $\text{SO}_2$  and 867 mbars of  $\text{CO}_2$  at a temperature of 343.6 K compared to various models



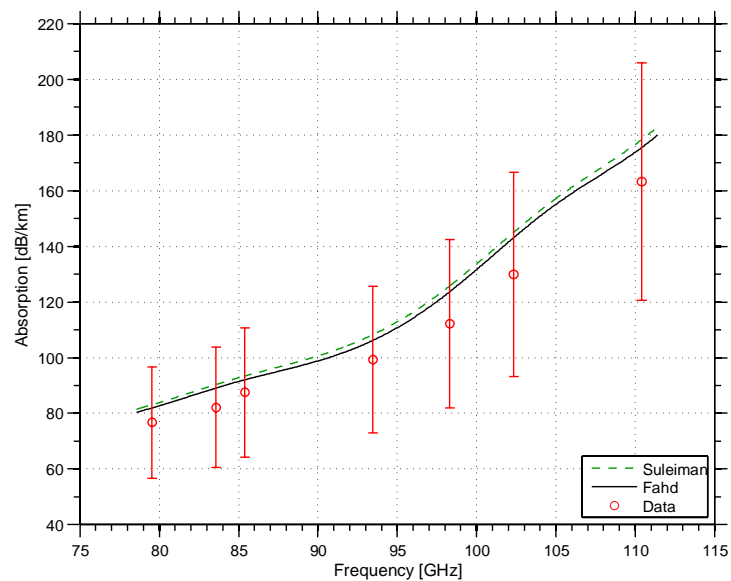


Figure 8: Opacity data using the 2.7-4 mm-wavelength system for a mixture of 51 mbars of  $\text{SO}_2$  and 1.953 bars of  $\text{CO}_2$  at a temperature of 343.6 K compared to various models

### 210 3.1. Suggested Model

211 Results indicate that the models for the millimeter-wavelength opacity  
 212 from  $\text{SO}_2$  in a  $\text{CO}_2$  atmosphere by Suleiman et al. (1996) and Fahd and  
 213 Steffes (1992) are both valid over the entire millimeter-wavelength range  
 214 under simulated conditions for the upper atmosphere of Venus. Previous  
 215 *centimeter-wavelength* laboratory measurements of  $\text{SO}_2$  in a  $\text{CO}_2$  atmosphere  
 216 were conducted by Steffes et al. (2015). Based on their data and the data  
 217 presented in this work, the suggested model is that from Fahd and Steffes  
 218 (1992), but using the updated  $\text{SO}_2$  line catalog from Pickett et al. (1998).  
 219 This model employs the Van Vleck-Weisskopf lineshape, and was developed  
 220 from measurements of  $\text{SO}_2/\text{CO}_2$  mixtures conducted at room temperature.  
 221 As per Fahd and Steffes (1992), we employ only the rotational line catalog to

222 compute opacity (JPL spectral line catalog, (Pickett et al., 1998)). Table 2  
 223 shows how both models perform relative to the centimeter- and millimeter-  
 224 wavelength laboratory data. Since most of the SO<sub>2</sub> absorption lines are  
 225 at millimeter-wavelengths, the millimeter-wavelength data fits the models  
 226 better.

227 When applied to a disk-averaged radiative transfer model similar to that  
 228 of Fahd and Steffes (1992), the difference in predicted emission using the two  
 229 models differs by 1.2 K at 8.7 GHz to 0.1 K at 100 GHz. While the resulting  
 230 difference is very small, this work has verified the accuracy of the frequency  
 231 dependence in the two models for SO<sub>2</sub> opacity. The only previous laboratory  
 232 measurement (Fahd and Steffes, 1992) measured the millimeter-wavelength  
 233 absorption of SO<sub>2</sub> at only a single wavelength (3.2 mm).

Table 2: The percentage of the measured data points within  $2\sigma$  uncertainty of the different models

SO <sub>2</sub> opacity model	Centimeter Wavelength (1.5-8 GHz) Steffes et al. (2015)	Millimeter Wavelength (80-150 GHz)	Total
Fahd and Steffes (1992)	82.97%	92.86%	85.88%
Suleiman (1997)	62.98%	88.10%	70.37%

#### 234 4. Conclusion

235 The objective of this work has been to advance the understanding of  
 236 the millimeter-wavelength properties of gaseous sulfur dioxide under Venus

237 conditions. Extensive laboratory measurements of the 2-4 mm-wavelength  
238 properties of sulfur dioxide under simulated upper troposphere conditions of  
239 Venus were conducted. These measurements along with previous laboratory  
240 measurements (Fahd and Steffes, 1992; Suleiman et al., 1996; Steffes et al.,  
241 2015) have been used to validate absorption formalisms across all microwave  
242 frequencies.

243 The formalism for sulfur dioxide opacity developed by Fahd and Steffes  
244 (1992) using the updated line catalog from Pickett et al. (1998) is able to fit  
245 85.88% of the laboratory data (centimeter-wavelength done by Steffes et al.  
246 (2015), millimeter-wavelength presented in this work) within  $2\sigma$  uncertainty.  
247 The bounds verified by this laboratory data are frequencies between 1-150  
248 GHz, temperatures between 307-550 K, and concentrations between 0-100%  
249  $\text{SO}_2$ /volume.

250 As a next step we are planning modifications to our laboratory system  
251 which will allow for measurement of the 2–4 millimeter-wavelength absorp-  
252 tion spectrum of gaseous  $\text{H}_2\text{SO}_4$  under simulated Venus conditions.

### 253 **Acknowledgement**

254 This work was supported by the NASA Planetary Atmospheres Program  
255 under Grant NNX11AD66G. We especially thanks Ms. Siri Anne Gomes  
256 and Dr. Doug King of Airgas USA for assistance in verifying gas mixture  
257 compositions.

### 258 **Appendix A. Measurement Theory**

259 The quality factor of a resonance is given by (Mattaei and Jones, 1980)

$$Q = \frac{2\pi f_0 \times \text{Energy Stored}}{\text{Average Power Loss}} \quad (\text{A.1})$$

260 where  $f_0$  is the resonant frequency. The Q of a resonance can be measured  
261 directly from  $f_0$  by dividing it by its half-power bandwidth (HPBW),

$$Q = \frac{f_0}{\text{HPBW}} \quad (\text{A.2})$$

262 The Q of a lossy gas ( $\epsilon'/\epsilon''$ ) and its opacity are related by

$$\alpha \approx \frac{\epsilon''\pi}{\epsilon'\lambda} = \frac{1}{Q_{gas}} \frac{\pi}{\lambda} \quad (\text{A.3})$$

263 where  $\epsilon'$  and  $\epsilon''$  are the real and imaginary permittivity of the gas,  $\lambda$  is the  
264 wavelength in km, and  $\alpha$  is the absorptivity of the gas in Nepers/km (1 Neper  
265 = 8.686 dB) (DeBoer and Steffes, 1994). Since Q is affected by more than  
266 just the gas under test, the Q of the gas-filled resonator is given by

$$\frac{1}{Q_{loaded}^m} = \frac{1}{Q_{gas}} + \frac{1}{Q_r} + \frac{1}{Q_{ext1}} + \frac{1}{Q_{ext2}} \quad (\text{A.4})$$

267 where  $Q_{loaded}^m$  is the measured quality factor of a resonance in the presence  
268 of a test gas,  $Q_{gas}$  is the quality factor of the gas under test,  $Q_r$  is the  
269 quality factor of the evacuated resonator in the absence of coupling losses,  
270 and  $Q_{ext1}$  and  $Q_{ext2}$  are the external coupling losses. Since the resonator used  
271 is symmetric, we assume  $Q_{ext1} = Q_{ext2}$ . Coupling losses can be derived from  
272 the transmissivity  $t = 10^{-S/10}$ , where  $S$  is the measured insertion loss of the  
273 resonator in decibels (dB) at the frequency of a particular resonance using  
274 the following relationship (Mattaei and Jones, 1980)

$$t = \left[ 2 \frac{Q^m}{Q_{ext}} \right]^2, \quad (\text{A.5})$$

$$Q_{ext} = \frac{2Q^m}{\sqrt{t}} \quad (\text{A.6})$$

275  $Q_r$  is related to the measured  $Q$  at a vacuum by

$$\frac{1}{Q_{vac}^m} = \frac{1}{Q_r} + \frac{1}{Q_{ext1}} + \frac{1}{Q_{ext2}} \quad (\text{A.7})$$

276 where  $Q_{vac}^m$  is the measured  $Q$  under vacuum conditions. Substituting equa-  
277 tion (A.6) into equations (A.4) and (A.7) gives

$$\frac{1}{Q_{gas}} = \frac{1 - \sqrt{t_{loaded}}}{Q_{loaded}^m} - \frac{1 - \sqrt{t_{vac}}}{Q_{vac}^m} \quad (\text{A.8})$$

278 where  $t_{loaded}$  and  $t_{vac}$  are the transmissivity of the resonance taken in loaded  
279 and vacuum conditions respectively. When gas is added to the resonator there  
280 is a shift in the center frequency corresponding to the refractive index of the  
281 test gas. Since the quality factor is reliant on the center frequency this will  
282 affect the comparison between the two measurements, even if the gas being  
283 tested is lossless. This effect is called dielectric loading (DeBoer and Steffes,  
284 1994). This effect can be corrected by performing additional measurements  
285 of the quality factor with a lossless gas present. Adding the lossless gas  
286 shifts the center frequency of the resonances, and by adding more or less gas  
287 the center frequency can be adjusted to be exactly the same as the lossy  
288 gas. These measurements are used in place of the vacuum measurements  
289 in equation (A.8) and by converting Nepers/km to dB/km equation (A.3)  
290 becomes

$$\alpha = 8.686 \frac{\pi}{\lambda} \left( \frac{1 - \sqrt{t_{loaded}}}{Q_{loaded}^m} - \frac{1 - \sqrt{t_{matched}}}{Q_{matched}^m} \right) \text{dB/km} \quad (\text{A.9})$$

291 An important note is that an absorption coefficient of 1 Nepers/km = 2  
 292 optical depths per km (or  $\text{km}^{-1}$ ) = 8.686 dB/km. The first notation is  
 293 used in electrical engineering, the second is the usual form in physics and  
 294 astronomy, and the third is the common (logarithmic) form. The third form  
 295 is often used in order to avoid a possible factor-of-two ambiguity.

296 It is also important to know the refractive index  $n_{ri}$  of the gas under test  
 297 since small changes in this value could alter the way electromagnetic waves  
 298 propagate through the medium. The refractive index of a gas is measured by  
 299 the frequency shift exhibited by the resonances and can be used to determine  
 300 the refractive index to a high level of accuracy. The refractivity ( $N$ ) of a gas  
 301 is given by

$$N = 10^6(n_{ri} - 1) \quad (\text{A.10})$$

302 The refractivity is measured by comparing the center frequency of the  
 303 resonator under gaseous conditions to the peak frequency of the resonances  
 304 at a vacuum and is give by

$$N = \frac{10^6(f_{vac} - f_{gas})}{f_{gas}} \quad (\text{A.11})$$

305 where  $N$  is the refractivity of the entire gas mixture  $f_{gas}$  is the peak frequency  
 306 of the resonance in presence of the test gas and  $f_{vac}$  the peak frequency of the  
 307 resonance at a vacuum (Tyler and Howard, 1969). The overall refractivity of  
 308 the mixture is the sum of the refractivity of the individual gases scaled with  
 309 their respective mole fractions.

## 310 **Appendix B. Measurement Uncertainties**

311 There are five sources of uncertainty for absorptivity measurements using  
 312 this system (Hanley and Steffes, 2007): instrumentation errors and electrical

313 noise ( $Err_{inst}$ ), errors in dielectric matching ( $Err_{diel}$ ), errors in transmissiv-  
 314 ity measurement ( $Err_{trans}$ ), errors due to resonance asymmetry ( $Err_{asym}$ ),  
 315 and errors in measurement conditions ( $Err_{cond}$ ) resulting from uncertainties  
 316 in temperature, pressure, and mixing ratio. The term  $Err$  represents  $2\sigma$   
 317 uncertainties.

318 Instrumental errors and electrical noise are due to the limited sensitivity  
 319 of the electrical devices and their ability to accurately measure bandwidth  
 320 ( $BW_{measured}$ ) and the center frequency ( $f_o$ ). Electrical noise arises from the  
 321 limited-stability frequency references and the noise of the internal electronics.  
 322 Electrical noise is uncorrelated and the best estimate of instrumental uncer-  
 323 tainty is the statistical variance of multiple measurements. The contribution  
 324 of electrical noise is outlined in Hanley et al. (2009),

$$Err_n = B \times \frac{S_n}{\sqrt{n_{samples}}} \quad (B.1)$$

325 where  $S_n$  is the sample standard deviation,  $B$  is the confidence coefficient  
 326 and  $n_{samples}$  is the number of independent measurements. For the millimeter-  
 327 wavelength system, five sets of independent measurements of each resonance  
 328 are taken. A confidence coefficient ( $B$ ) of 2.776 is used (for each 5 sample  
 329 data point). This corresponds to the 95% confidence interval (approximately  
 330  $2\sigma$ ) (Student, 1908). The center frequency standard deviation is very small  
 331 and its effect on the uncertainty in  $Q$  is negligible. Therefore,  $S_n$  is the  
 332 standard deviation of the bandwidth of the measurements.

333 The HP 8564E spectrum analyzer is used for measuring the resonances  
 334 in the millimeter-wavelength system. Its manufacturer-specified instrumen-  
 335 tal uncertainties are  $3\sigma$  values (Hewlett-Packard, 1997). The  $3\sigma$  standard  
 336 deviation for the center frequency and bandwidth are estimated by

$$Err_o \leq \pm(f_o \times f_{ref\ acc} + 0.05 \times SPAN + 0.15 \times RBW + 10)(Hz) \quad (B.2)$$

337

$$Err_{BW} \leq \pm(BW_{measured} \times f_{ref\ acc} + 4 \times N_H + 2 \times LSD)(Hz) \quad (B.3)$$

338 where  $f_{ref\ acc}$  is given as

$$f_{ref\ acc} = (\text{aging} \times \text{time since calibration}) + \text{inital achievable accuracy} \\ + \text{temperature stability} \quad (B.4)$$

339 and  $f_o$ , SPAN, RBW,  $N_H$ , and LSD are the center frequency, frequency  
 340 span, resolution bandwidth, harmonic number, and least significant digit of  
 341 the bandwidth measurement, respectively. LSD is calculated as

$$LSD = 10^x \quad (B.5)$$

342 where  $x$  is the the smallest positive integer value of  $x$  such that SPAN <  
 343  $10^{x+4}$ . For SPAN  $\leq 2$  MHz  $\times N_h$ , Equation (B.2) becomes

$$Err_o \leq \pm(f_o \times f_{ref\ acc} + 0.01 \times SPAN + 0.15 \times RBW + 10)(Hz) \quad (B.6)$$

344 For the spectrum analyzer used,  $f_{ref\ acc}$  reduces to

$$f_{ref\ acc} = (10^{-7} \times \text{years since calibrated}) + 3.2 \times 10^{-8} \quad (B.7)$$

345 The worst case scenario is used to transform the uncertainty in center  
 346 frequency and bandwidth for both loaded and dielectrically matched mea-  
 347 surements into an uncertainty in absorptivity as described in DeBoer and  
 348 Steffes (1994).



$$Err_{\Psi}^2 = \langle F_l^2 \rangle + \langle F_m^2 \rangle - \langle F_l F_m \rangle \quad (\text{B.8})$$

349 where

$$\langle F_i^2 \rangle = \frac{\Upsilon_i^2}{f_{oi}^2} \left[ \frac{Err_o^2}{Q_l^2} + Err_{BW}^2 + Err_{Ni}^2 + \frac{2Err_o Err_{BW}}{Q_i} \right], i = l, m \quad (\text{B.9})$$

350

$$\langle F_l F_m \rangle = -\frac{\Upsilon_l \Upsilon_m}{f_{ol} f_{om}} \left[ \frac{Err_o^2}{Q_i Q_m} + Err_{BW}^2 + \frac{Err_o Err_{BW}}{Q_l} + \frac{Err_o Err_{BW}}{Q_m} \right] \quad (\text{B.10})$$

351

$$Q_i = \frac{f_{oi}}{f_{BW_i}}, i = l, m \quad (\text{B.11})$$

352

$$\Upsilon_i = 1 - \sqrt{t}, i = l, m \quad (\text{B.12})$$

353 where  $l, m$  denote loaded and dielectrically matched cases respectively and  
 354  $f_{ol,om}$  and  $f_{BWl,BWm}$  represent center frequency and bandwidth of loaded and  
 355 dielectrically matched cases respectively. The  $2\sigma$  uncertainty of the measured  
 356 gas absorption due to instrumental errors and electrical noise is given by

$$Err_{inst} = \pm \frac{8.686\pi}{\lambda} Err_{\Psi} \text{ (dB/km)} \quad (\text{B.13})$$

357 where  $\lambda$  is the wavelength in km.

358 Errors in dielectric matching occur when the center frequency of the  
 359 matched measurements are not precisely aligned with the center frequency  
 360 of the loaded measurement. Since the  $Q$  of the resonator can vary slightly,  
 361 this causes an uncertainty in the  $Q$  of the matched measurement at the true  
 362 center frequency of the loaded measurement. The method used to calculate  
 363 the magnitude of this effect is similar to Hanley and Steffes (2007). While  
 364 this error is the smallest due to the high precision of the software controlled  
 365 matching, it is important to calculate and account for. The magnitude of this

366 effect is calculated by comparing the  $Q$  of the three vacuum measurements  
 367 to that of the dielectric matched measurements

$$\left(\frac{dQ}{df}\right)_i = \left| \frac{Q_{vac,i} - Q_{matched,i}}{f_{vac,i} - f_{matched,i}} \right| \text{ for } i = 1, 2, 3 \quad (\text{B.14})$$

368 The maximum of the three values is used to calculate a  $dQ$  value

$$dQ = \left(\frac{dQ}{df}\right)_{max} \times |f_{loaded} - f_{matched}| \quad (\text{B.15})$$

369 where  $f_{loaded}$  and  $f_{matched}$  are the center frequencies of the resonances under  
 370 loaded and matched conditions. The error in absorptivity due to imperfect  
 371 dielectric matching is then computed by propagating  $\pm dQ$  through Equation  
 372 (A.9).

$$\begin{aligned} Err_{diel} &= \frac{8.686\pi}{\lambda} \\ &\times \left| \left( \frac{1 - \sqrt{t_{loaded}}}{Q_{loaded}^m} - \frac{1 - \sqrt{t_{matched}}}{Q_{matched}^m + dQ} \right) - \left( \frac{1 - \sqrt{t_{loaded}}}{Q_{loaded}^m} - \frac{1 - \sqrt{t_{matched}}}{Q_{matched}^m - dQ} \right) \right| \\ &(\text{dB/km}) \end{aligned} \quad (\text{B.16})$$

373 Transmissivity errors are due to the uncertainties in the measurement  
 374 amplitude. This is caused by variations in gains or losses of the millimeter-  
 375 wavelength instruments (signal generators and spectrum analyzer), cables,  
 376 adapters, and waveguides used in this system. This is done by taking multiple  
 377 test measurements of signal loss through the system without the FPR and  
 378 finding the standard deviation ( $S_N$ ) of the signal loss and weighing it by its  
 379 confidence coefficient

$$Err_{msl} = \frac{4.303}{\sqrt{3}} S_N \quad (\text{B.17})$$

380 For the millimeter-wavelength system, the signal level measurements in-  
 381 volve sampling the RF power with a WR-10 20 dB directional coupler to feed  
 382 the harmonic mixer for down-conversion and detection. While this ensures  
 383 that the input to the harmonic mixer does not exceed its maximum allowed  
 384 input power of -10 dBm, the WR-10 20dB directional coupler does not uni-  
 385 formly sample the input signal throughout the entire frequency range. To  
 386 compensate for this, an additional 1.5 dB uncertainty is added to insertion  
 387 loss error. The signal generator has a temperature stability of 1 dB/10° C,  
 388 but an internal temperature equilibrium is reached after two hours (Hewlett-  
 389 Packard, 1997). Since the instruments are operated at a constant tempera-  
 390 ture this uncertainty can be disregarded. The total uncertainty in insertion  
 391 loss for the millimeter-wavelength system is calculated by

$$Err_{ins\ loss} = Err_{msl} + 1.5 \text{ (dB)} \quad (\text{B.18})$$

392 The error in insertion loss is used to compute the transmissivity error

$$Err_{t,i} = \frac{1}{2} (10^{-S_i - Err_{ins\ loss}} - 10^{-S_i + Err_{ins\ loss}}), i = l, m \quad (\text{B.19})$$

393 where l, m are the loaded and matched cases, respectively, and S is the  
 394 insertion loss of the resonator. This is used to compute the  $2\sigma$  uncertainties  
 395 in opacity and is expressed as

$$Err_{trans} = \frac{8.686\pi}{2\lambda} \times \left| \left( \frac{\sqrt{t_l + Err_{t,l}} - \sqrt{t_l - Err_{t,l}}}{Q_{loaded}^m} - \frac{\sqrt{t_m - Err_{t,m}} - \sqrt{t_m + Err_{t,m}}}{Q_{matched}^m} \right) \right|$$

(dB/km).

$$(\text{B.20})$$

396 Errors from asymmetry are due to the asymmetric nature of the reso-  
 397 nances. These are more prominent at low temperatures and short wave-  
 398 lengths. Errors due to the asymmetry result from the disproportionate asym-  
 399 metric broadening of the loaded measurements compared to the matched  
 400 measurements. Equivalent full bandwidths based on assuming symmetry of  
 401 the high and low sides of the resonances are calculated as

$$BW_{high} = 2 \times (f_{high} - f_{center}) \quad (B.21)$$

402

$$BW_{low} = 2 \times (f_{center} - f_{low}) \quad (B.22)$$

403 where  $BW_{high}$ ,  $BW_{low}$ ,  $f_{high}$ ,  $f_{center}$ , and  $f_{low}$  are the high bandwidth, low  
 404 bandwidth, higher frequency half power point, center frequency, and lower  
 405 frequency half power point, respectively. For a perfectly symmetric reso-  
 406 nance,  $BW_{high} = BW_{low}$ . The difference between the opacities calculated  
 407 using  $BW_{high}$  and  $BW_{low}$  is defined as  $Err_{asym}$  and is calculated by

$$Err_{asym} = \frac{8.686\pi}{\lambda} \times \left| \left( \frac{1 - \sqrt{t_{loaded}}}{Q_{loaded,high}^m} - \frac{1 - \sqrt{t_{matched}}}{Q_{matched,high}^m} \right) - \left( \frac{1 - \sqrt{t_{loaded}}}{Q_{loaded,low}^m} - \frac{1 - \sqrt{t_{matched}}}{Q_{matched,low}^m} \right) \right|$$

(dB/km)

(B.23)

408 where  $Q_{matched,high/low}^m$  and  $Q_{loaded,high/low}^m$  are the measured Q's evaluated  
 409 using the high and low bandwidths for loaded and matched cases.

410 The uncertainties in measured temperature, pressure, and concentra-  
 411 tion in the millimeter-wavelength system contribute to the total uncertainty  
 412 due to the measurement conditions ( $Err_{cond}$ ). While uncertainties in mea-

413 surement conditions do not directly affect the measurements of millimeter-  
 414 wavelength absorptivity, they still need to be accounted for when evaluating  
 415 the opacity formalisms. It is computed by

$$Err_{cond} = \sqrt{Err_{temp}^2 + Err_p^2 + Err_c^2} \text{ (dB/km)} \quad (\text{B.24})$$

416 with  $Err_{temp}$ ,  $Err_p$ , and  $Err_c$  representing the  $2\sigma$  uncertainties in temper-  
 417 ature, pressure, and concentration (or mole fraction) respectively. Each of  
 418 these are calculated by taking the maximum modeled opacity with each un-  
 419 certainty minus the minimum modeled opacity and halving the difference.

420 Since  $Err_{cond}$  is dependent on the opacity model, this uncertainty is main-  
 421 tained separately from  $Err_{tot}$ . Thus the total 95% confidence for the mea-  
 422 surement uncertainty is expressed in dB/km as per Hanley and Steffes (2007)

$$Err_{tot} = \sqrt{Err_n^2 + Err_{diel}^2 + Err_{trans}^2 + Err_{asym}^2} \text{ (dB/km)}. \quad (\text{B.25})$$

#### 423 *Accuracy of Constituents*

424 It is necessary to ensure that the gases used in each experiment are cor-  
 425 rectly characterized. The CO<sub>2</sub> used was Ultra High Purity (UHP) grade.  
 426 The test gas was comprised of 84.7% SO<sub>2</sub> and 15.3% N<sub>2</sub>. In our experiments,  
 427 the partial pressure of N<sub>2</sub> never exceeds 20 mbar. While including N<sub>2</sub> allows  
 428 for a simulation similar to that of the Venus atmosphere, it does not affect  
 429 the outcome of our models.

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We measure the millimeter absorption of SO<sub>2</sub> in a Venus atmosphere.

We compare these measurements with previously used formalism.

We determine which formalism fits better fits the measured data.

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